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# Flowfield Comparisons: Tunnel 9 Case 3745

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# Tunnel 9 – Run 3745

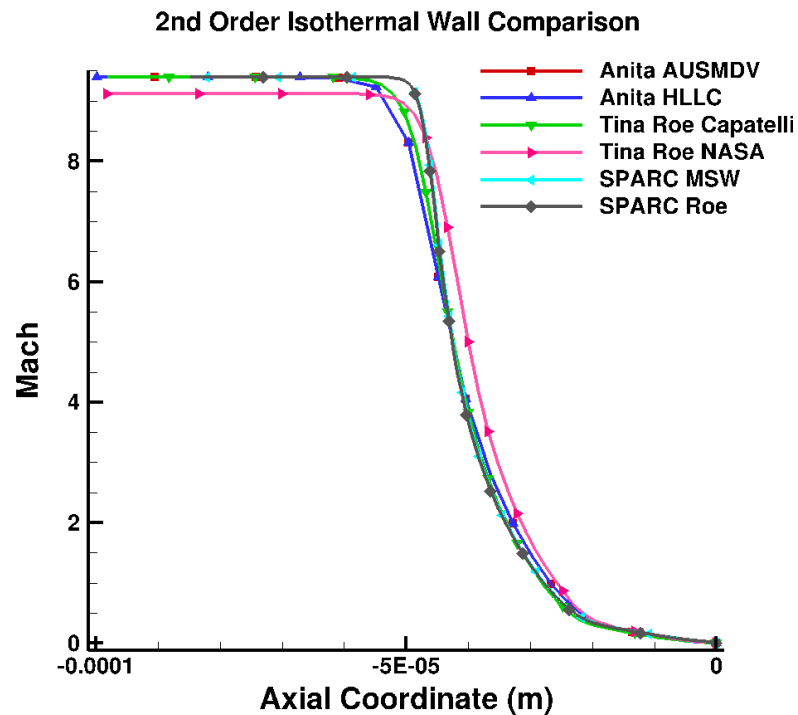
- Conditions and geometry from AIAA-2014-3108
  - $T_\infty = 54.42 \text{ K}$ ;  $\rho_\infty = 4.861123e - 3 \text{ kg/m}^3$ ;  $U_\infty = 1413.083 \text{ m/s}$
  - Sphere cone  $7^\circ$  half-angle, nose radius 0.006 inches



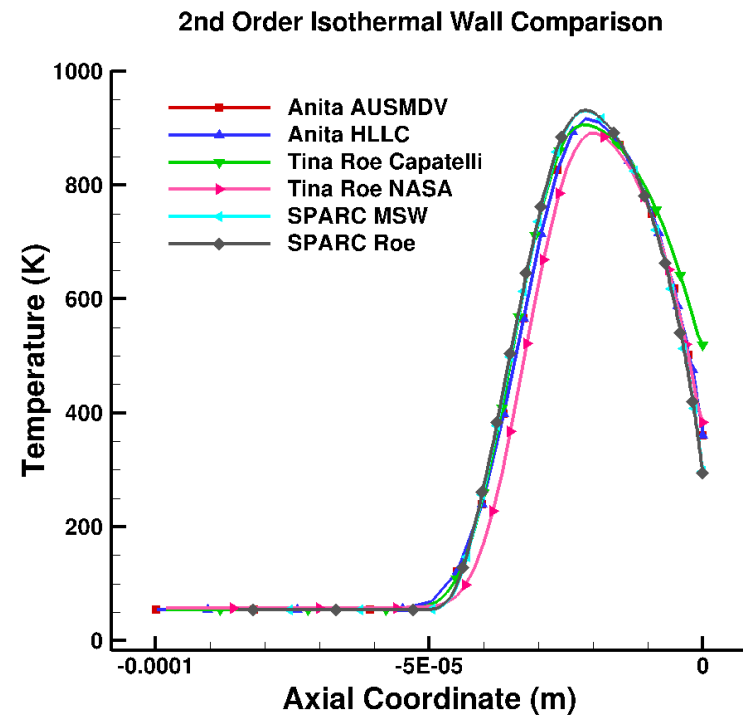
- Sandia Grid
  - 20 cells on the nose; 800 on the body; 200 cells wall-normal
  - Manually shock tailored
- Sandia Computations made with SPARC
  - Cell-centered finite volume using Modified Steger-Warming and Roe inviscid flux schemes
  - Minmod limiter using MUSCL TVD scheme
  - Viscous fluxes evaluated using weighted-least squares gradients

# Stagnation Line Comparisons: 2<sup>nd</sup> Order

## Mach Number

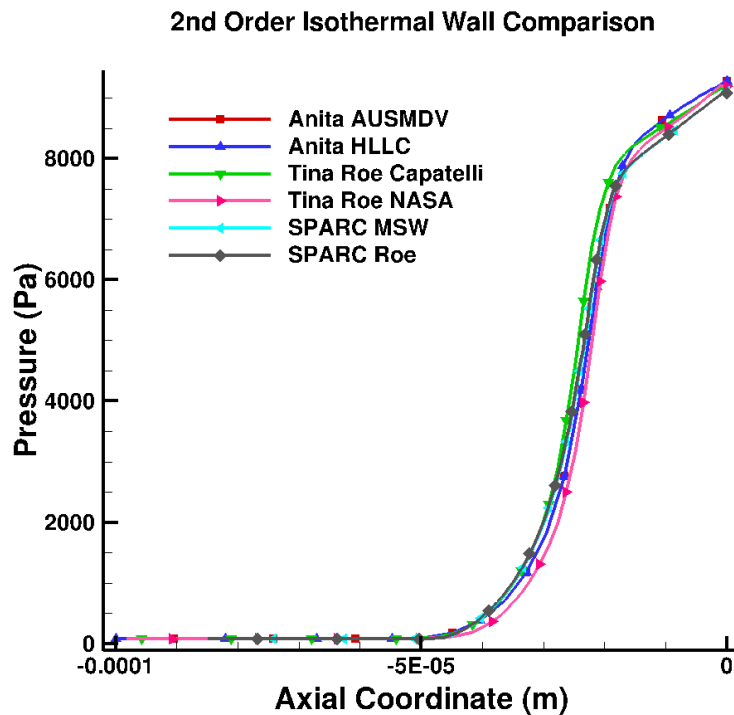


## Temperature

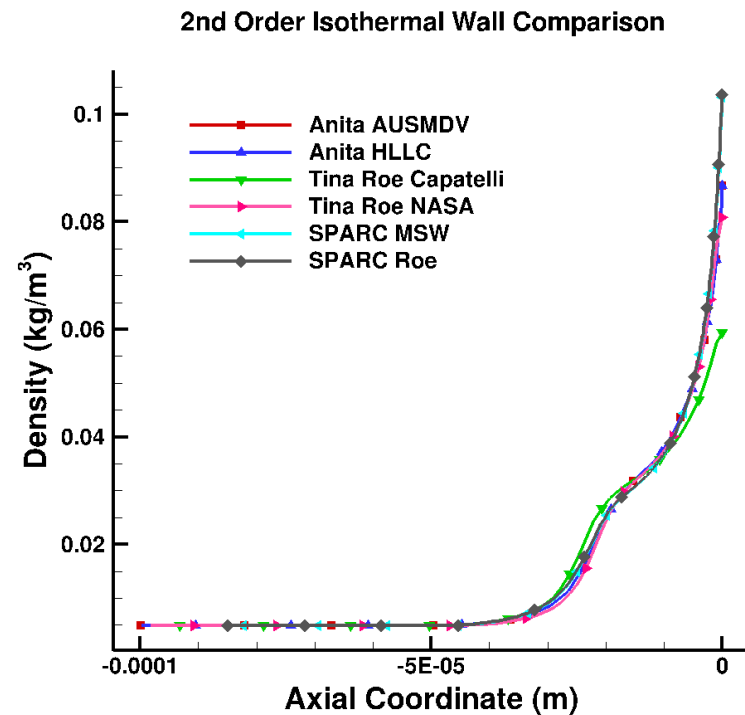


# Stagnation Line Comparisons: 2<sup>nd</sup> Order

## Pressure

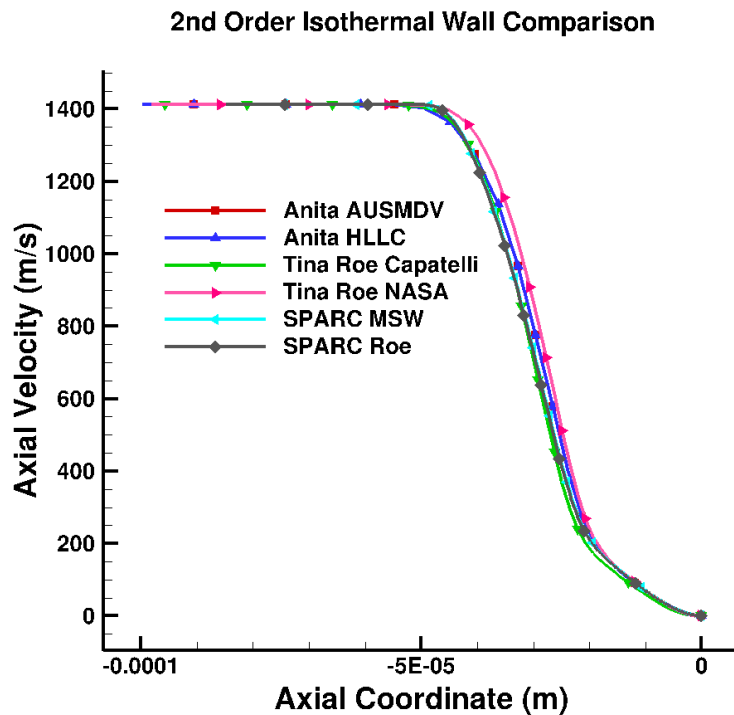


## Density



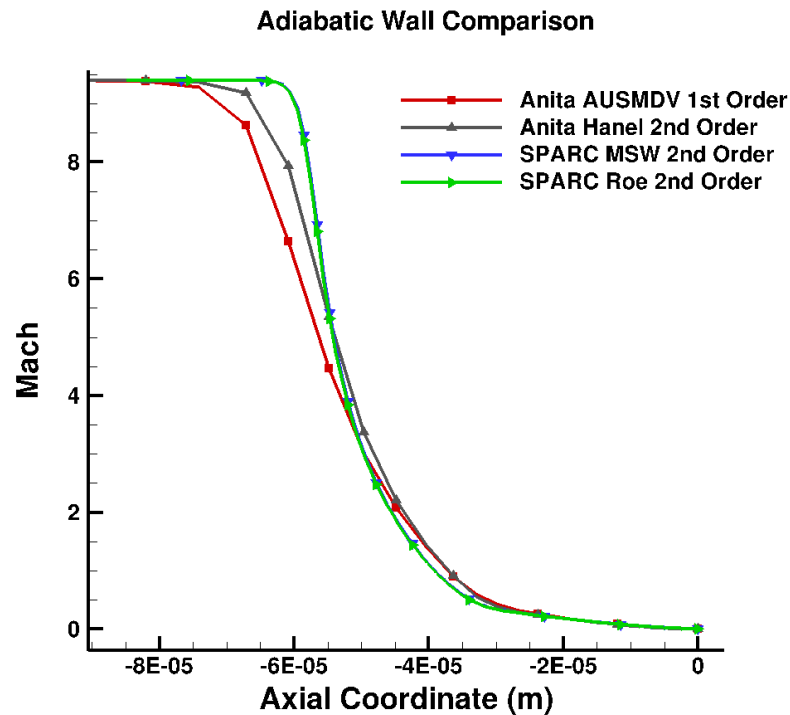
# Stagnation Line Comparisons: 2<sup>nd</sup> Order

## Velocity

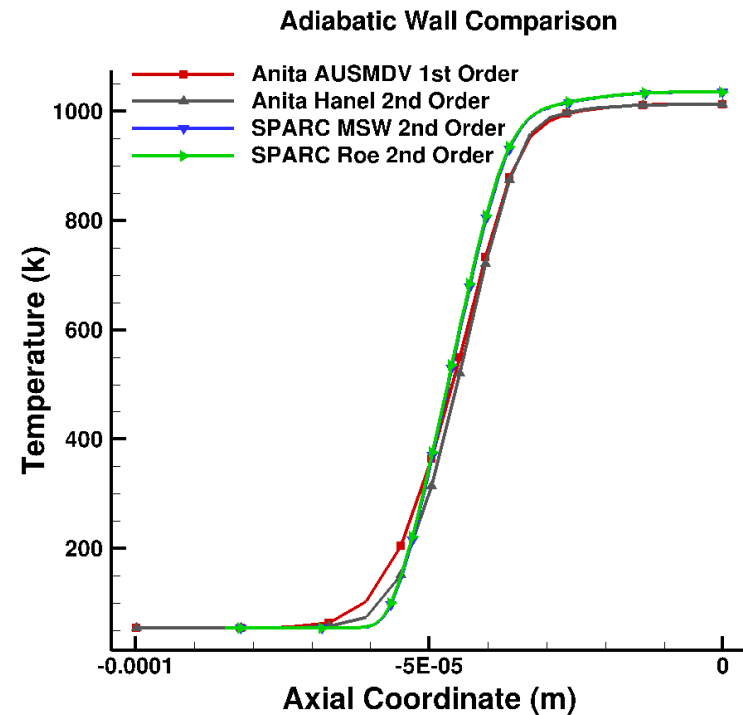


# Stagnation Line Comparisons: Adiabatic

## Mach Number

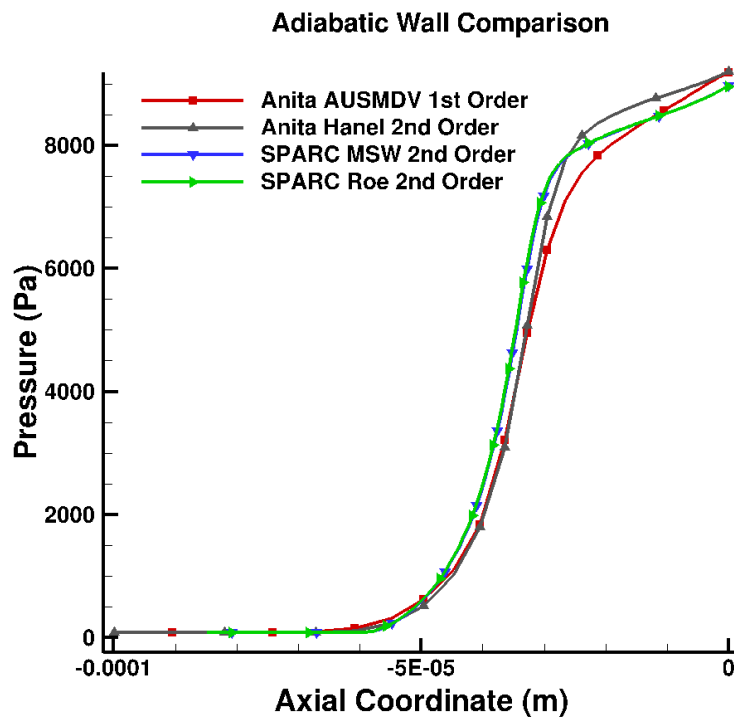


## Temperature

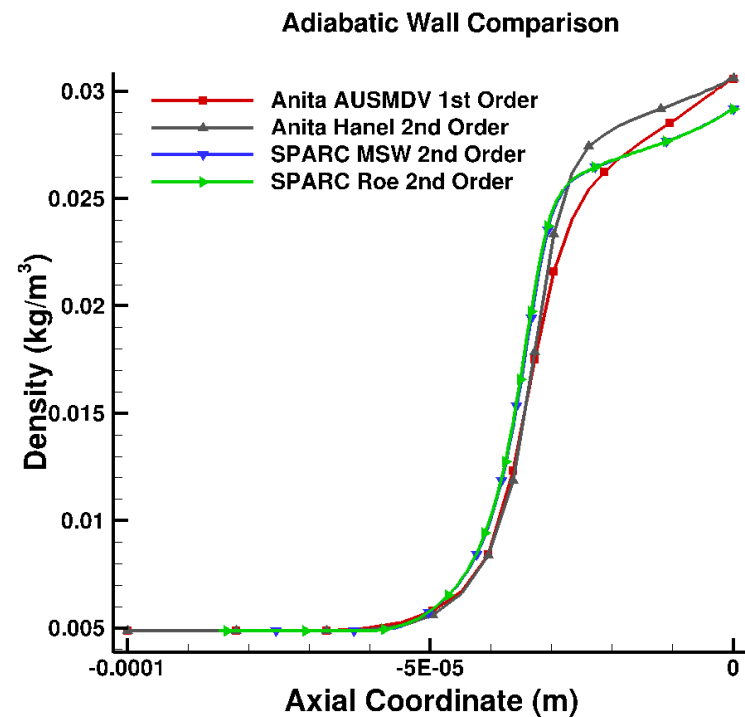


# Stagnation Line Comparisons: Adiabatic

## Pressure

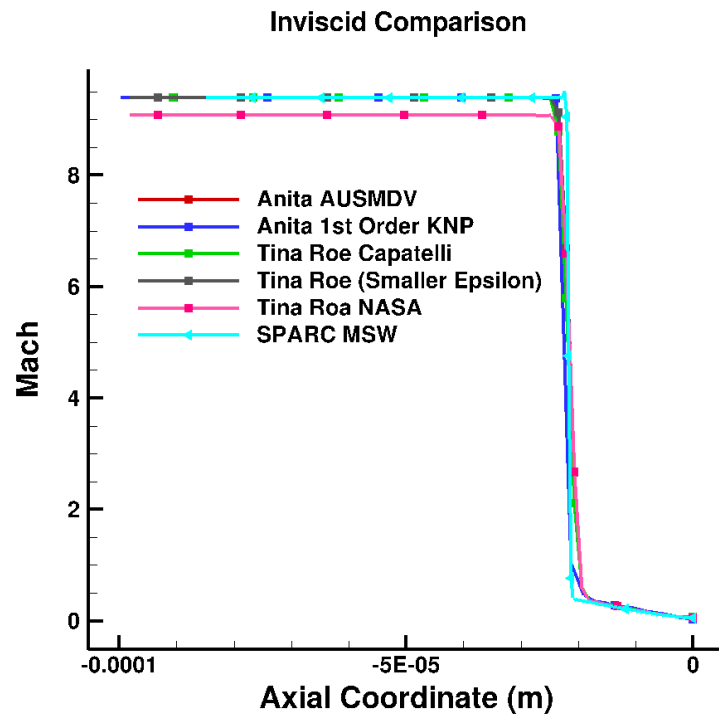


## Density



# Stagnation Line Comparisons: Inviscid

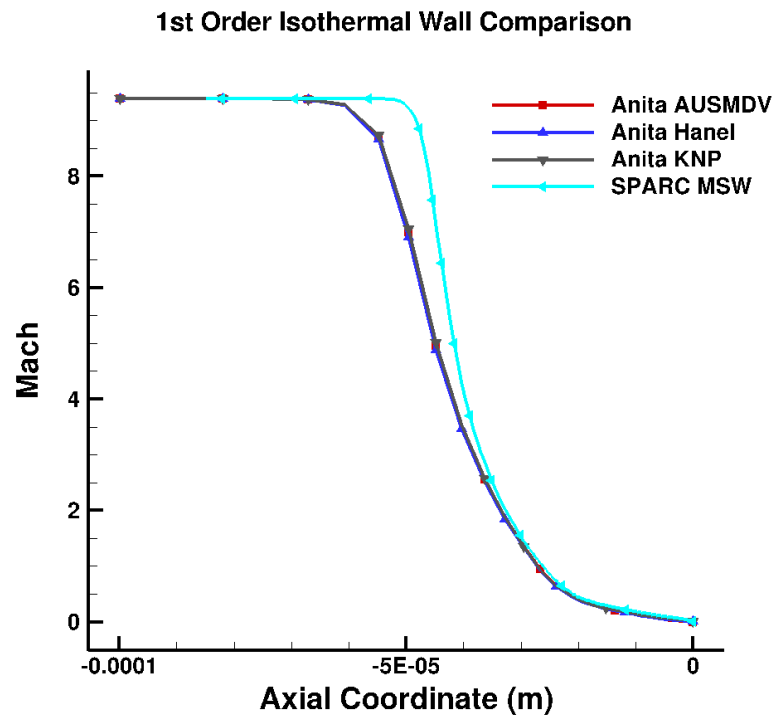
## Mach Number



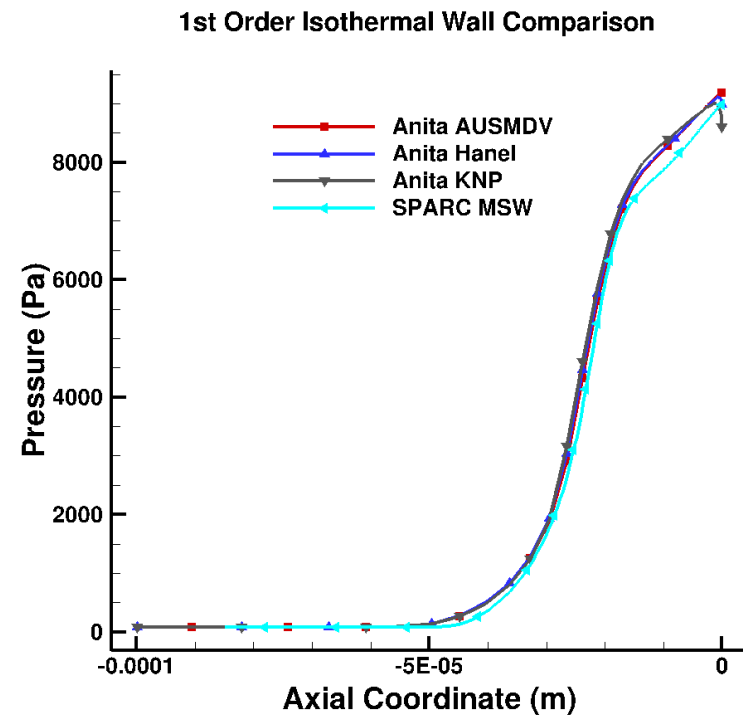


# Stagnation Line Comparisons: 1<sup>st</sup> Order

## Mach Number

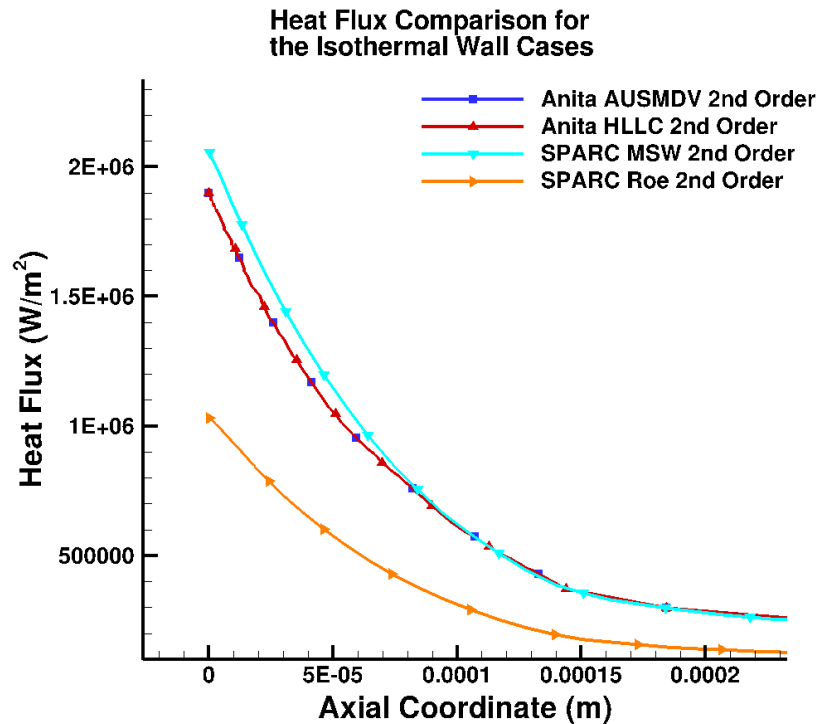


## Pressure

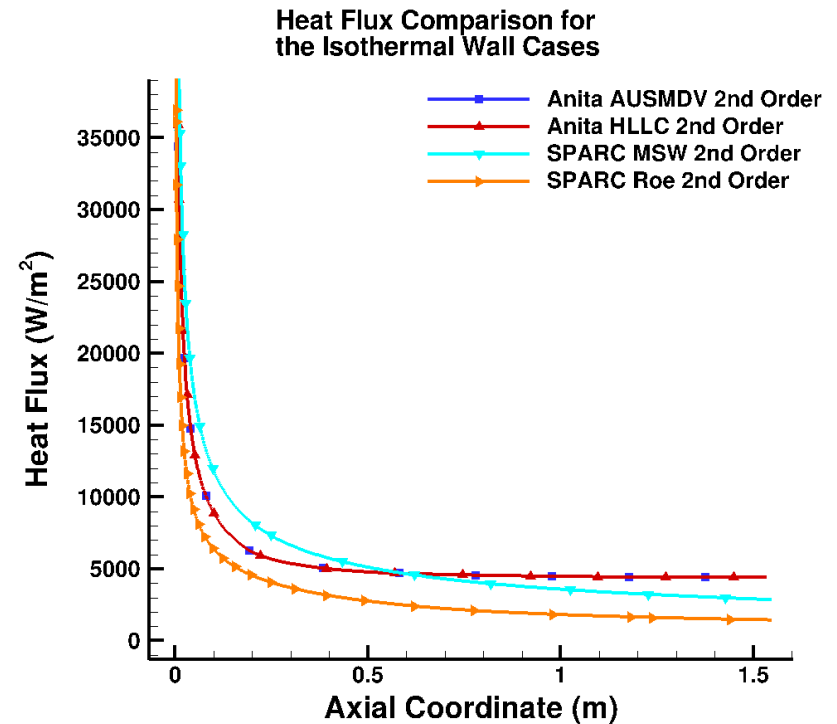


# Surface Heat Flux Comparisons: 2<sup>nd</sup> Order

## Nose

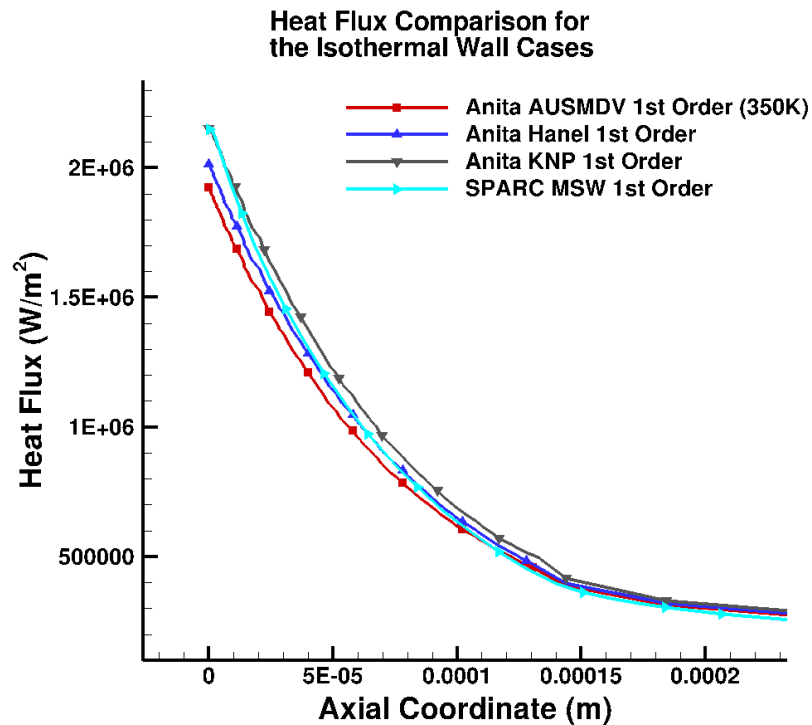


## Full Vehicle

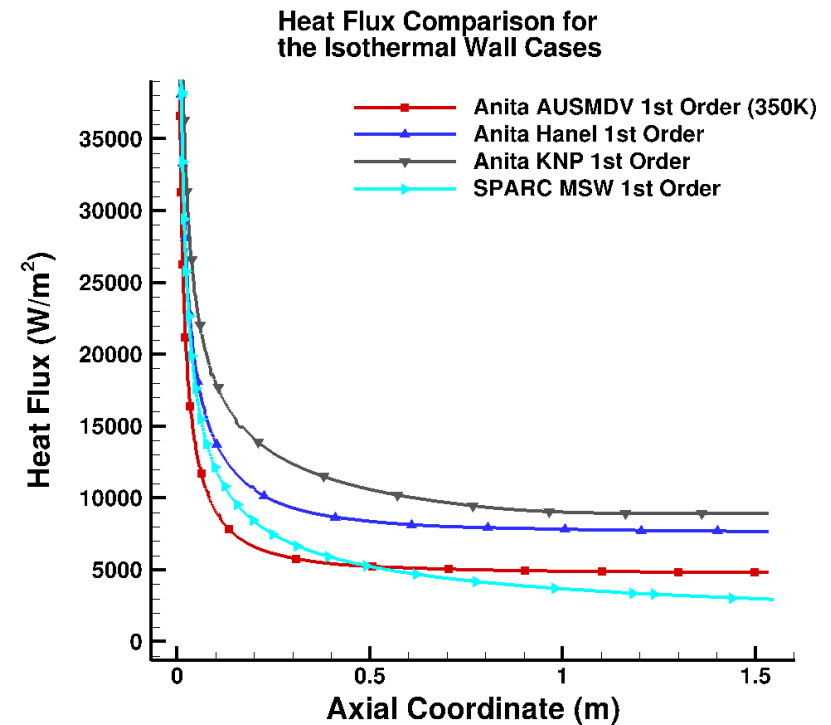


# Surface Heat Flux Comparisons: 1st Order

## Nose



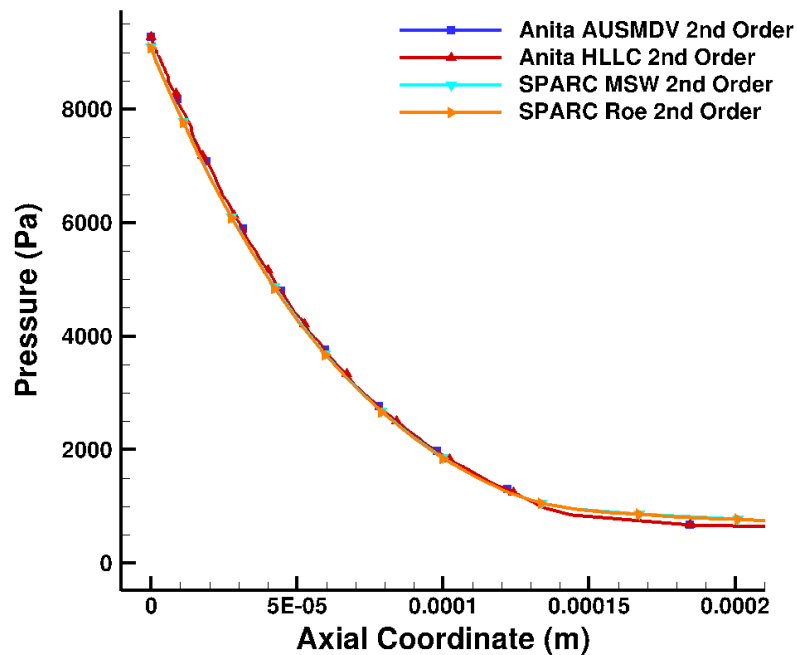
## Full Vehicle



# Surface Pressure Comparisons: 2<sup>nd</sup> Order

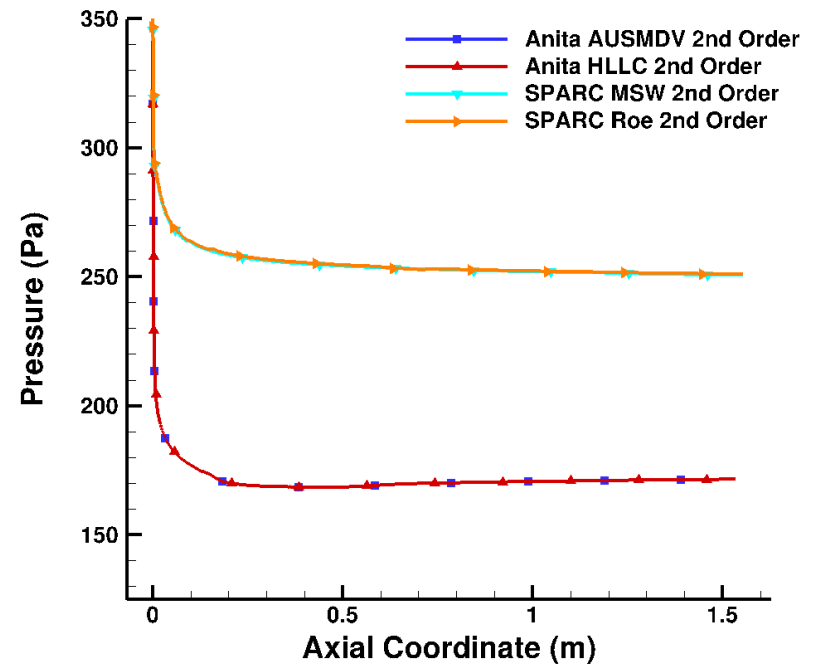
## Nose

Surface Pressure Comparison for  
the Isothermal Wall Cases



## Full Vehicle

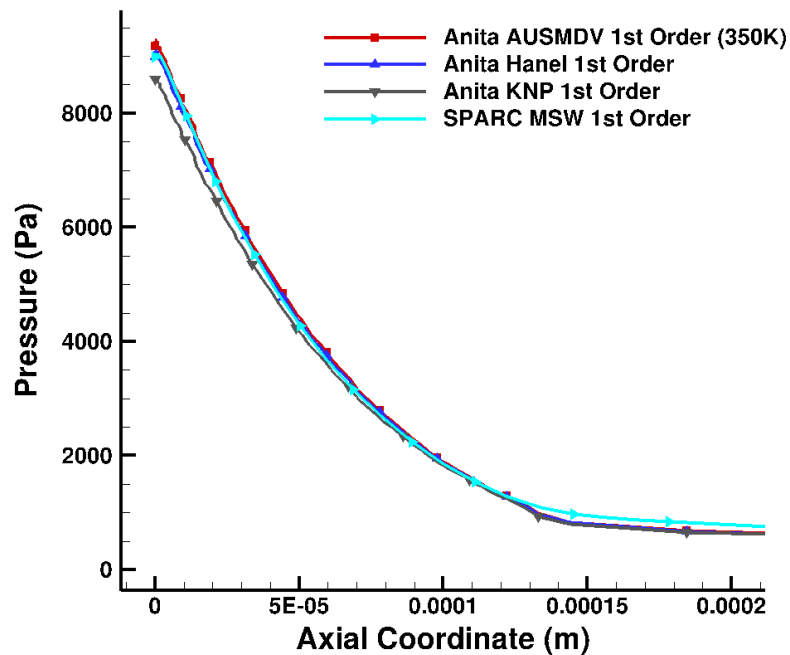
Surface Pressure Comparison for  
the Isothermal Wall Cases



# Surface Pressure Comparisons: 1st Order

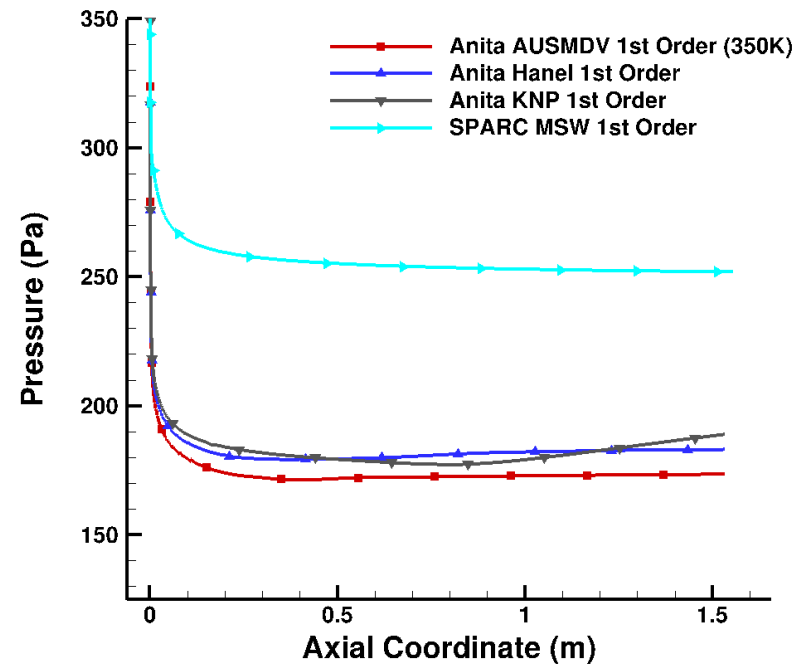
## Nose

Surface Pressure Comparison for  
the Isothermal Wall Cases



## Full Vehicle

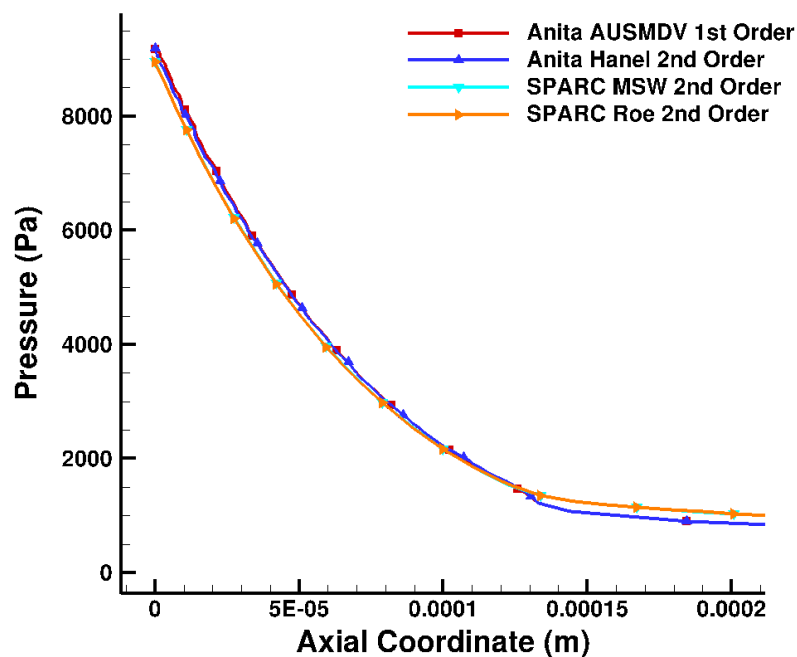
Surface Pressure Comparison for  
the Isothermal Wall Cases



# Surface Pressure Comparisons: Adiabatic

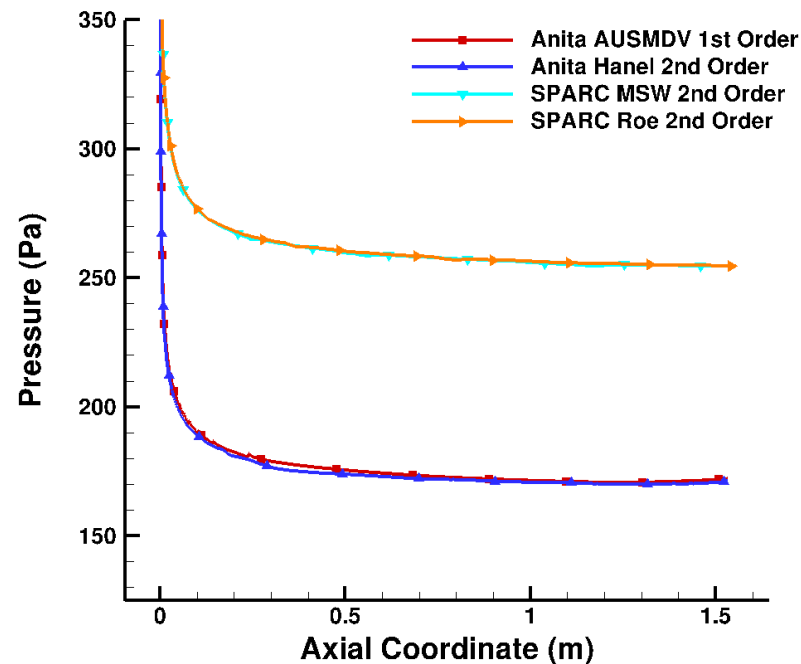
## Nose

Surface Pressure Comparison for  
the Adiabatic Wall Cases



## Full Vehicle

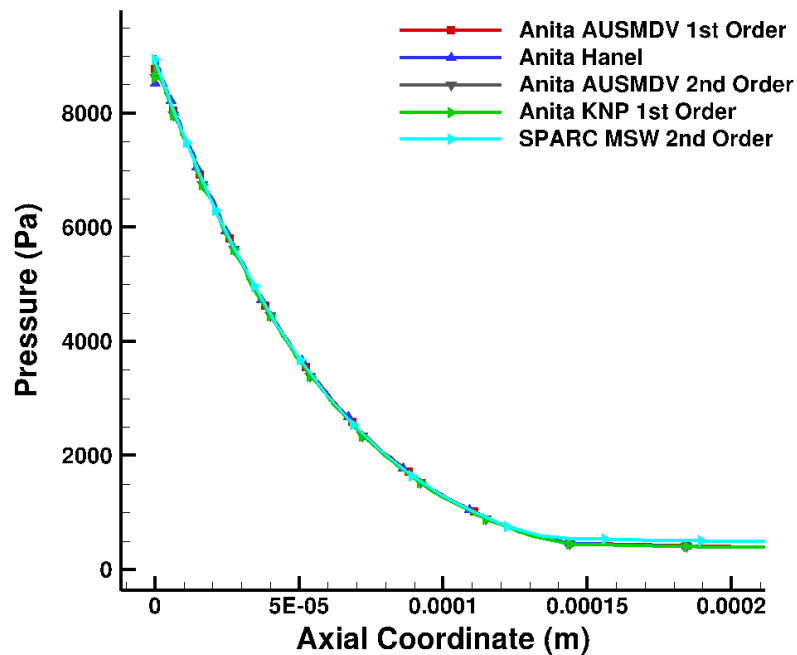
Surface Pressure Comparison for  
the Adiabatic Wall Cases



# Surface Pressure Comparisons: Inviscid

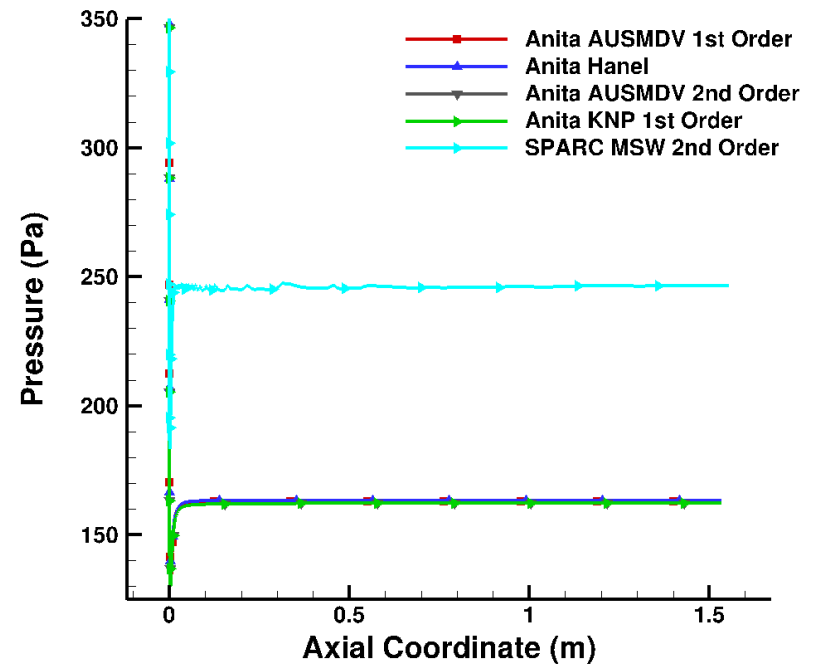
## Nose

Surface Pressure Comparison for the Inviscid Cases



## Full Vehicle

Surface Pressure Comparison for the Inviscid Cases



# Initial Conclusions

- Stagnation line comparisons
  - All codes are very similar – outliers are Tina Roe Capatelli and Tina Roe NASA (density and Mach plots)
- Surface comparisons
  - Nose comparisons are very similar – outlier SPARC Roe
  - Full vehicle comparisons show differences
  - Heat flux
    - SPARC MSW and SPARC Roe distinctly different
    - SPARC has slightly different heating trend compared to Anita
  - Pressure
    - All codes predict similar pressure on the nose
    - Frustrum pressures within each code are all very similar
    - Between codes the pressure differs significantly
- Wind tunnel data update
  - Data originally marked as export controlled and are currently being reviewed for release.